Advanced nonlinear structural analysis



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#### **Outline**

- Sources of Nonlinearities
- Introduction to Nonlinear Structural
   Analysis
- Finite element formulation for nonlinear dynamic analysis
- Advanced topics









#### **Sources of nonlinearity**

- ☐ Geometric
- Material
- ☐ Force Boundary Conditions
- ☐ Displacement Boundary Conditions

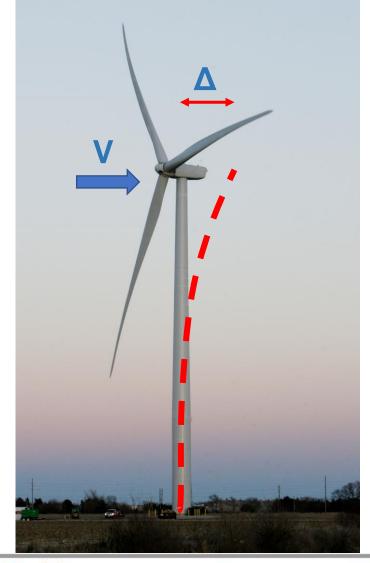






#### SECOND-ORDER **ELASTIC** FIRST-ORDER. **ELASTIC** ELASTIC BIFURCATION . FIRST-ORDER > RIGID PLASTIC FIRST-ORDER Load **ELASTIC-PLASTIC HINGE** Lateral SECOND-ORDER SECOND-ORDER **ELASTIC-PLASTIC HINGE** RIGID-PLASTIC SECOND-ORDER DISTRIBUTED PLASTICITY Lateral Deflection -- $\Delta$

#### Pushover analysis of wind turbine towers









Methods		Constitutive Relationship	Features Equilibrium Formulation	Geometric Compatibility
First-order	Elastic Rigid–plastic Elastic–plastic hinge Distributed plasticity	Elastic Rigid plastic Elastic perfectly plastic Inelastic	Original undeformed geometry	Small strain and small displacement
Second-order	Elastic Rigid–plastic Elastic–plastic hinge Distributed plasticity	Elastic Rigid–plastic Elastic perfectly plastic Inelastic	Deformed structural geometry ( $P$ - $\Delta$ and $P$ - $\delta$ )	Small strain and moderate rotation (displacement may be large)
True large displacement	Elastic Inelastic	Elastic Inelastic	Deformed structural geometry	Large strain and large deformation

#### **Structural Analysis Methods**







ļ.	$\Delta P$	Workshop 2: Study of structural and foundation systems of Wind Turbines  Features				
-						
	$-\frac{1}{2}$ $\delta$	Constitutive Relationship	Equilibrium Formulation	Geometric Compatibility		
P	1 ,	Elastic Rigid plastic Elastic perfectly plastic Inelastic	Original undeformed geometry	Small strain and small displacement		
Second-order	Elastic Rigid–plastic Elastic–plastic hinge Distributed plasticity	Elastic Rigid–plastic Elastic perfectly plastic Inelastic	Deformed structural geometry ( $P$ - $\Delta$ and $P$ - $\delta$ )	Small strain and moderate rotation (displacement may be large)		
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#### **Structural Analysis Methods**







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#### **Structural Analysis Methods**







## **General Guidelines**

First-order analysis VS. second-order analysis

True large displacement analysis

Elastic analysis VS. Inelastic analysis

Bowing effect / Wagner effect / Other effects

Steel nonlinearity

Concrete nonlinearity

Composite materials nonlinearity

Other nonlinearities







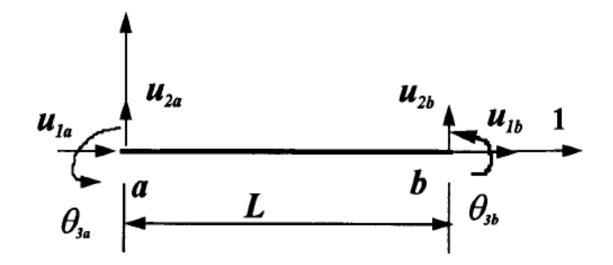


#### **Geometric Nonlinearity Formulation**

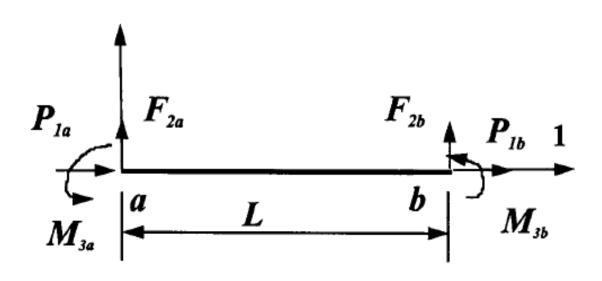
$$\{F\} = [K]\{D\}$$

$$\{F\} = \{P_{1a}, F_{2a}, M_{3a}, P_{1b}, F_{2b}, M_{3b}\}^T$$

$$\{\boldsymbol{D}\} = \{u_{1a}, u_{2a}, \theta_{3a}, u_{1b}, u_{2b}, \theta_{3b}\}^T$$







### **Geometric Nonlinearity Formulation**

$$\{F\} = [K]\{D\}$$
$$[K] = [K_e] + [K_g]$$

$$\begin{bmatrix} \frac{AE}{L} & 0 & 0 & -\frac{AE}{L} & 0 & 0 \\ \frac{12EI}{L^3} \phi_1 & \frac{-6EI}{L^2} \phi_2 & 0 & \frac{-12EI}{L^3} \phi & \frac{-6EI}{L^2} \phi_2 \\ 4\phi_3 & 0 & \frac{6EI}{L^2} \phi_2 & 2\phi_4 \\ \frac{AE}{L} & 0 & 0 \\ \frac{12EI}{L^3} \phi & \frac{6EI}{L^2} \phi_2 \\ 4\phi_3 & 4\phi_3 \end{bmatrix}$$

$$\frac{1 + \sum_{n=1}^{\infty} \frac{1}{(2n+1)!} \left[ \mp (kL)^{2} \right]^{n}}{12\phi}$$

$$\frac{1}{2} + \sum_{n=1}^{\infty} \frac{1}{(2n+2)!} \left[ \mp (kL)^2 \right]^n$$

$$\phi_{3} \qquad \frac{\frac{1}{3} + \sum_{n=1}^{\infty} \frac{2(n+1)}{(2n+3)!} \left[ \mp (kL)^{2} \right]^{n}}{4\phi}$$

$$\phi_4 \qquad \frac{\frac{1}{6} + \sum_{n=1}^{\infty} \frac{1}{(2n+3)!} \left[ \mp (kL)^2 \right]^n}{2\phi}$$

$$\phi \qquad \frac{1}{12} + \sum_{k=1}^{\infty} \frac{2(n+1)}{(2n+4)!} \left[ \mp (kL)^2 \right]^n$$

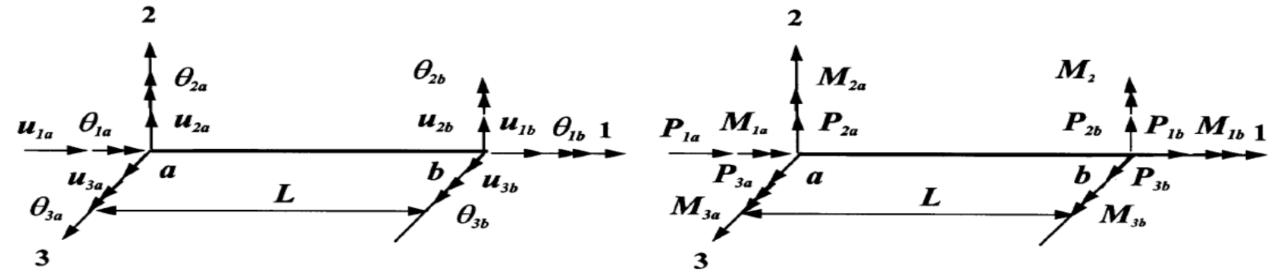
#### **Geometric Nonlinearity Formulation**

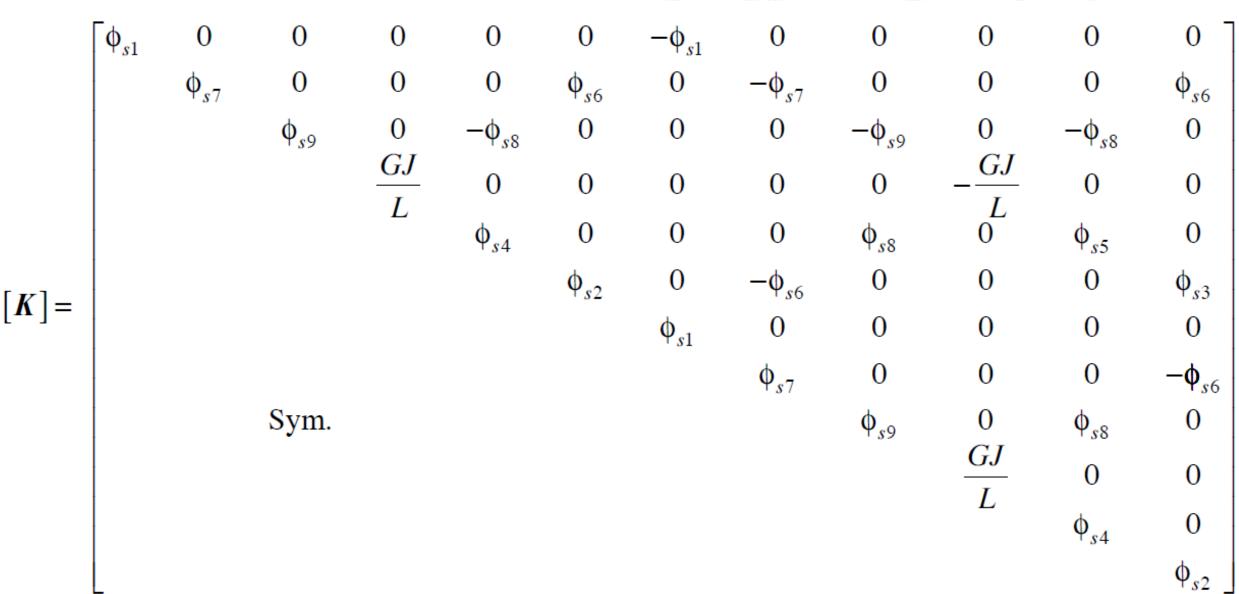
$$\{F\} = [K]\{D\}$$



$$\{F\} = \{P_{1a}, F_{2a}, F_{3a}, M_{1a}, M_{2a}, M_{3a}, P_{1b}, F_{2b}, F_{3b}, M_{1b}, M_{2b}, M_{3b}\}^{T}$$

$$\{\boldsymbol{D}\} = \{u_{1a}, u_{2a}, u_{3a}, \theta_{1a}, \theta_{2a}, \theta_{3a}, u_{1b}, u_{2b}, u_{3b}, \theta_{1b}, \theta_{2b}, \theta_{3b}\}^T$$





#### **3-D Stiffness matrix**









Linear Elastic Matrix

Stability Functions  $S_i$ 

$$\frac{1-\frac{E}{4P}$$

$$\frac{1}{1 - \frac{EA}{4P^3L^2} \left[ H_y + H_z \right]}$$

Compression

 $(\alpha L)(\sin \alpha L - \alpha L \cos \alpha L)$ 

 $(\alpha L)(\alpha L - \sin \alpha L)$ 

 $2\phi_{\alpha}$ 

 $(\beta L)(\sin \beta L - \beta L \cos \beta L)$ 

 $(\beta L)(\beta L - \sin \beta L)$ 

 $2\phi_{\rm g}$ 

 $(\alpha L)^2 (1 - \cos \alpha L)$ 

 $(\alpha L)^3 \sin \alpha L$ 

 $12\phi_{\alpha}$ 

 $(\beta L)^2 (1 - \cos \beta L)$ 

 $(\beta L)^3 \sin \beta L$ 

 $12\phi_{\rm g}$ 

 $2-2\cos\alpha L-\alpha L\sin\alpha L$ 

 $2-2\cos\beta L-\beta L\sin\beta L$ 

 $\phi_{si}$ 

 $\phi_{s1} = S_1 \frac{EA}{I}$ 

 $\phi_{s2} = S_2 \frac{(4 + \phi_y)EI_Z}{(1 + \phi_y)L}$ 

 $\phi_{s3} = S_2 \frac{(2 - \phi_y)EI_Z}{(1 + \phi_{...})L}$ 

 $\phi_{s4} = S_4 \frac{(4 + \phi_z)EI_y}{(1 + \phi_z)L}$ 

 $\phi_{s5} = S_2 \frac{(2 - \phi_z)EI_y}{(1 + \phi_z)L}$ 

 $\phi_{s6} = S_6 \frac{6EI_Z}{(1+\phi_{sa})L^2}$ 

 $\phi_{s7} = S_7 \frac{12EI_Z}{(1 + \phi_w)L^3}$ 

 $\phi_{s8} = S_8 \frac{6EI_y}{(1+\phi_s)L^2}$ 

 $\phi_{s9} = S_9 \frac{12EI_y}{(1+\phi_0)L^3}$ 

 $\alpha = \sqrt{P/EI_z}$ 

 $\beta = \sqrt{P/EI_{y}}$ 

 $S_3$ 

 $S_6$ 

 $S_7$ 

 $S_8$ 

 $S_9$ 

 $\phi_{\alpha}$ 

$$\frac{1}{1 - \frac{EA}{4P^3L^2} \left[ H_y' + H_z' \right]}$$

Tension

$$\frac{EA}{4P^3L^2}\left[H_y' + H_z'\right]$$

$$1 - \frac{EA}{4P^3L^2} \left[ H_y' + H_z' \right]$$

 $(\beta L)(\beta L \cosh \beta L - \sinh \beta L)$ 

 $(\beta L)(\sinh \beta L - \beta L)$ 

 $2\phi_{\rm B}$ 

 $(\alpha L)^2 (\cosh \alpha L - 1)$ 

 $6\phi_{\alpha}$ 

 $(\alpha L)^3 \sinh \alpha L$ 

 $12\phi_{\alpha}$ 

 $(\beta L)^2 (\cosh \beta L - 1)$ 

 $(\beta L)^3 \sinh \beta L$ 

 $12\phi_{\rm g}$ 

 $2 - 2\cosh\alpha L + \alpha L \sinh\alpha L$ 

 $2 - 2 \cosh \beta L + \beta L \sinh \beta L$ 

$$\frac{1}{4P^3L^2} \frac{[H_y + H_z]}{[H_y + H_z]}$$

$$L)(\alpha L \cosh \alpha L - \sinh \alpha L)$$

$$\frac{(\alpha L)(\alpha L \cosh \alpha L - \sinh \alpha L)}{(\alpha L)(\alpha L \cosh \alpha L - \sinh \alpha L)}$$

$$\frac{4PL}{L} = \frac{1}{2}$$

$$L)(\alpha L \cosh \alpha L - \sinh \alpha L)$$

$$4P^{3}L^{2}L \qquad \qquad ^{2}J$$

$$L)(\alpha L \cosh \alpha L - \sinh \alpha L)$$

$$\frac{4PL}{(\alpha L \cosh \alpha L - \sinh \alpha L)}$$

$$\frac{L}{L}(\alpha L \cosh \alpha L - \sinh \alpha L)$$

$$4P^{3}L^{2}L^{-y}$$

$$L)(\alpha L \cosh \alpha L - \sinh \alpha L)$$

$$4P^{3}L^{2} [^{11}y + ^{12}z]$$

$$(\alpha L \cosh \alpha L - \sinh \alpha L)$$

$$1 - \frac{1}{4P^3L^2} \left[ H_y + H_z \right]$$

$$(\alpha L \cosh \alpha L - \sinh \alpha L)$$

$$(\alpha L \cosh \alpha L - \sinh \alpha L)$$

$$)(\alpha L \cosh \alpha L - \sinh \alpha L)$$

$$)(\alpha L \cosh \alpha L - \sinh \alpha L)$$

$$(\alpha L \cosh \alpha L - \sinh \alpha L)$$

$$(\alpha L \cosh \alpha L - \sinh \alpha L)$$

$$-\frac{1}{4P^3L^2} \left[ \Pi_y + \Pi_z \right]$$

$$(\alpha L \cosh \alpha L - \sinh \alpha L)$$

$$4P^{3}L^{2}L^{-y} \qquad ^{2}$$

$$(\alpha L \cosh \alpha L - \sinh \alpha L)$$

$$-\frac{1}{4P^3L^2}[H_y + H_z]$$

$$\frac{1}{4P^3L^2}\left[H_y' + H_z'\right]$$

 $(\alpha L)(\sinh \alpha L - \alpha L)$ 

 $\phi_{e1} = \frac{AE}{I}; \qquad \phi_{e2} = \frac{4EI_Z}{I}$ 

 $\phi_{e3} = \frac{2EI_z}{I}; \quad \phi_{e4} = \frac{4EI_y}{I}$ 

 $\phi_{e5} = \frac{2EI_y}{I}$ ;  $\phi_{e6} = \frac{6EI_Z}{I^2}$ 

 $\phi_{g14} = \frac{M_{yb}}{I}; \quad \phi_{g15} = \frac{M_{za}}{I}; \quad \phi_{g16} = \frac{M_{zb}}{I}$ 

Geometric Nonlinear Matrix

 $\phi_{g7} = \phi_{g9} = \frac{6F_{xb}}{5I}; \quad \phi_{g6} = \phi_{g8} = \frac{F_{xb}}{10}; \quad \phi_{g10} = \frac{M_{za} + M_{zb}}{I^2}$ 

 $\phi_{g1} = 0$ ;  $\phi_{g2} = \phi_{g4} = \frac{2F_{xb}L}{15}$ ;  $\phi_{g3} = \phi_{g5} = \frac{F_{xb}L}{30}$ 

 $\phi_{g11} = \frac{M_{ya} + M_{yb}}{I^2}; \quad \phi_{g12} = \frac{M_{ya}}{I}; \quad \phi_{g13} = \frac{M_{xb}}{I}$ 

 $\phi_{g17} = \frac{F_{xb}I_p}{AI}$ ;  $\phi_{g18} = \frac{M_{zb}}{6} - \frac{M_{za}}{3}$ ;  $\phi_{g19} = \frac{M_{ya}}{3} - \frac{M_{yb}}{6}$  $\phi_{g20} = \frac{M_{za} + M_{zyb}}{6}; \quad \phi_{g21} = \frac{M_{ya} + M_{yb}}{6}$ 

 $\phi_{e7} = \frac{12EI_Z}{r^3}$ ;  $\phi_{e8} = \frac{6EI_y}{r^2}$ 

 $\phi_{e9} = \frac{12EI_y}{I^3}$ 

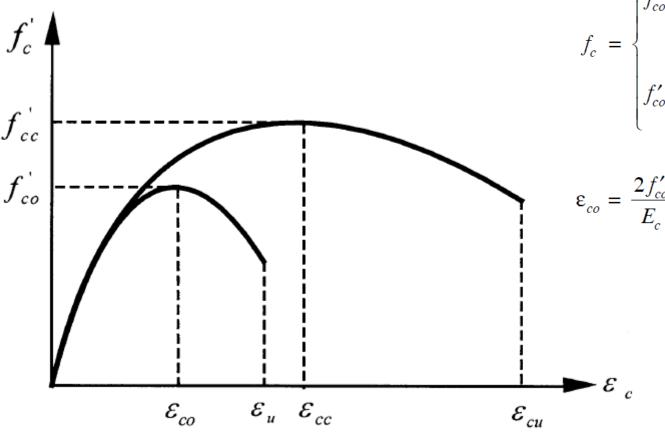
DAAD

$$H_{y} = \beta L (M_{ya}^{2} + M_{yb}^{2}) (\cot \beta L + \beta L \cos ec^{2}\beta L) - 2(M_{ya} + M_{yb})^{2} + 2\beta L M_{ya} M_{yb} (\cos ec\beta L) (1 + \beta L \cot \beta L)$$

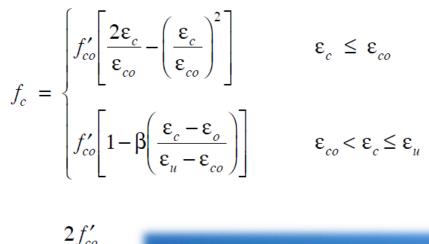
$$\begin{split} H_z &= \alpha L (M_{za}^2 + M_{zb}^2) (\cot \alpha L + \alpha L \cos ec^2 \alpha L) - 2 (M_{za} + M_{zb})^2 + 2\alpha L M_{za} M_{zb} (\cos ec \alpha L) (1 + \alpha L \cot \alpha L) \\ H_y' &= \beta L (M_{ya}^2 + M_{yb}^2) (\coth \beta L + \beta L \cos ech^2 \beta L) - 2 (M_{ya} + M_{yb})^2 + 2\beta L M_{ya} M_{yb} (\cos ech \beta L) (1 + \beta L \coth \beta L) \end{split}$$

 $H'_{z} = \alpha L (M_{za}^2 + M_{zb}^2) (\coth \alpha L + \alpha L \cos e ch^2 \alpha L) - 2(M_{za} + M_{zb})^2 + 2\alpha L M_{za} M_{zbb} (\cos e ch \alpha L) (1 + \alpha L \coth \alpha L)$ 

#### **Material Nonlinearity Formulations**



Idealized stress-strain curves for concrete in uniaxial compression





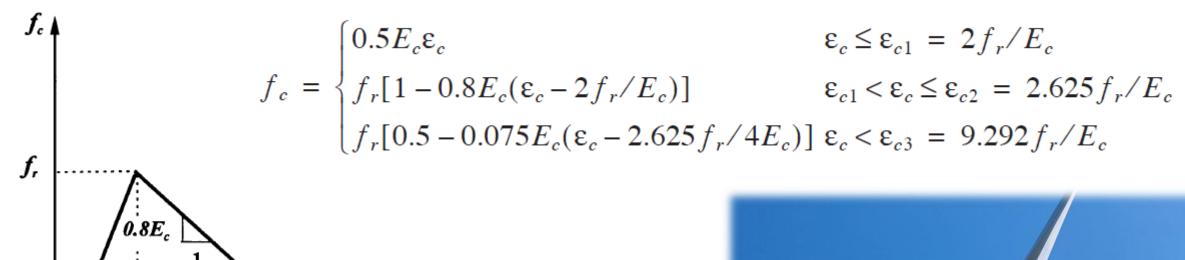
**Concrete: towers / foundation / piles** 







#### **Material Nonlinearity Formulations**



Idealized stress-strain curves for concrete in uniaxial tension



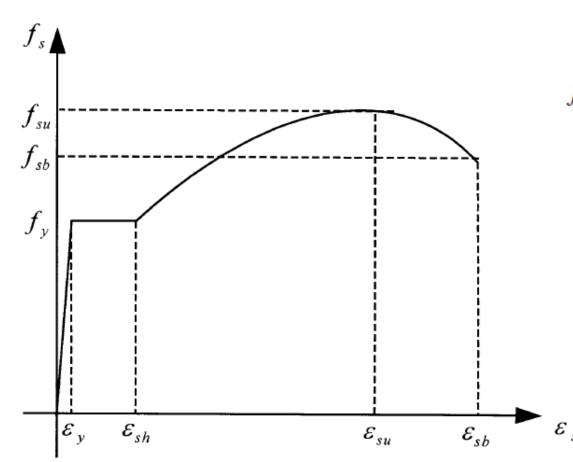








#### **Material Nonlinearity Formulations**



$$f_{s} = \begin{cases} E_{s} \varepsilon_{s} & 0 \leq \varepsilon_{s} \leq \varepsilon_{y} \\ f_{y} & \varepsilon_{sy} < \varepsilon_{s} \leq \varepsilon_{sh} \\ f_{y} + \frac{\varepsilon_{s} - \varepsilon_{sh}}{\varepsilon_{su} - \varepsilon_{sh}} (f_{su} - f_{y}) & \varepsilon_{sh} < \varepsilon_{s} \leq \varepsilon_{su} \\ f_{u} - \frac{\varepsilon_{s} - \varepsilon_{su}}{\varepsilon_{sb} - \varepsilon_{su}} (f_{su} - f_{sb}) & \varepsilon_{cu} < \varepsilon_{s} \leq \varepsilon_{sb} \end{cases}$$



Idealized stress-strain curves for steel in uniaxial tension & compression

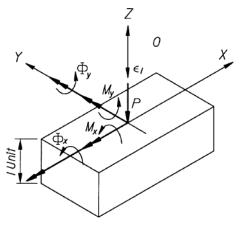
Steel: towers / reinforcement / shaft / hub etc. DAAD Deutscher Akademischer German Academic Exchar



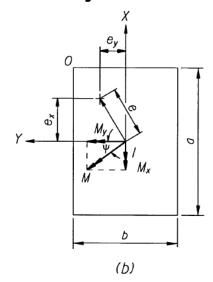


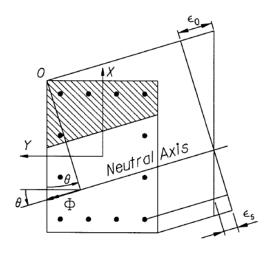


#### **Nonlinear Section Analysis**

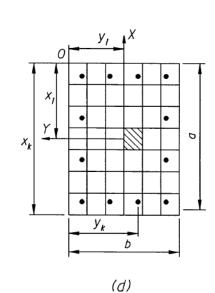


(a)





(c)



#### $\phi_x = \epsilon / y$

#### **Compatibility equations**

$$\phi_{v} = \varepsilon / x$$

$$P = \int_{A} \sigma \, dA = \sum_{i=1}^{n} \sigma_{i} A_{i}$$

Equilibrium equations 
$$M_x = \int_A \sigma y dA = \sum_{i=1}^n \sigma_i y_i A_i$$

$$M_{y} = \int_{A} \sigma x \, dA = \sum_{i=1}^{n} \sigma_{i} x_{i} A_{i}$$

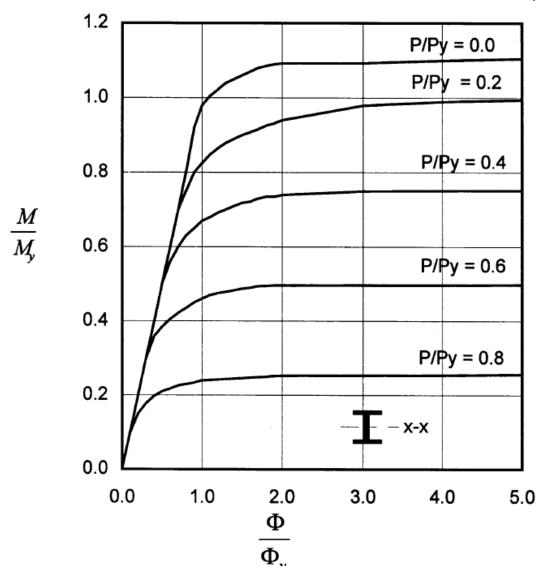


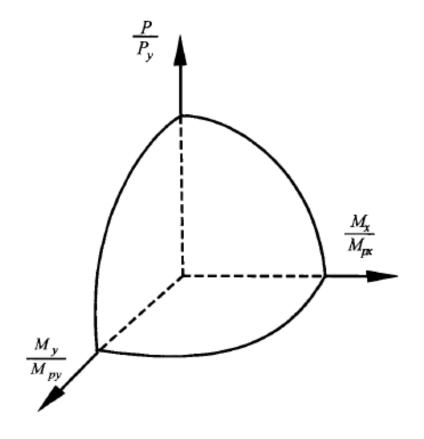




#### **Nonlinear Section Analysis**

Moment—thrust—curvature curve for a steel I-section (2-D loads)



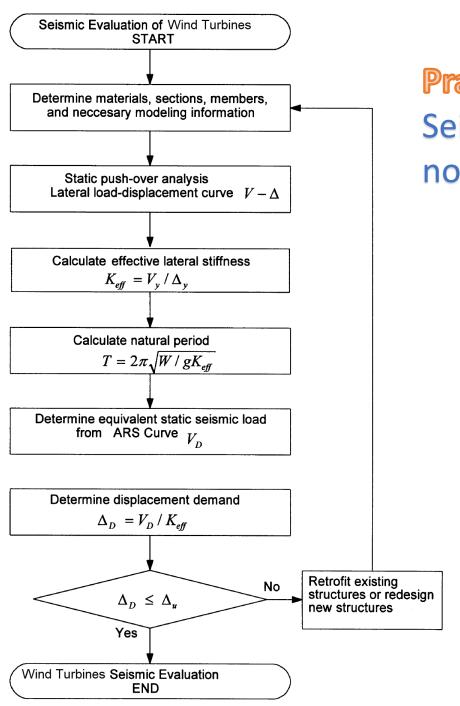


General yield surface for a steel I-section (3-D loads)

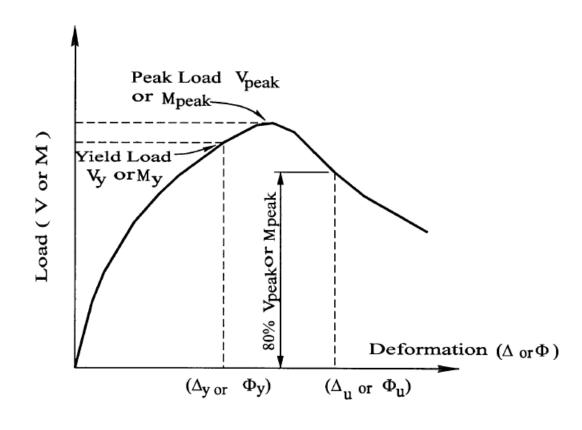








Seismic evaluation of wind turbines with nonlinear behavior









#### **Linear** and Nonlinear Dynamic Analysis

$$[M] \cdot \ddot{u} + [C] \cdot \dot{u} + [K] \cdot u = R(t)$$

|M | Mass matrix

C Damping matrix

[K] Stiffness matrix (constant)

R(t) Vector of external loads

 $u,\dot{u},\ddot{u}$  Displacement, velocity and acceleration vectors

 $\left[ M \right]$  ,  $\left[ C \right]$  ,  $\left[ K \right]$  symmetrical







#### Linear and **Nonlinear** Dynamic Analysis

$$[M] \cdot \ddot{u} + [C] \cdot \dot{u} + [K(t)] \cdot u = R(t)$$

Mass matrix

C Damping matrix

[K(t)] Stiffness matrix (time depended)

R(t) Vector of external loads

 $u,\dot{u},\ddot{u}$  Displacement, velocity and acceleration vectors

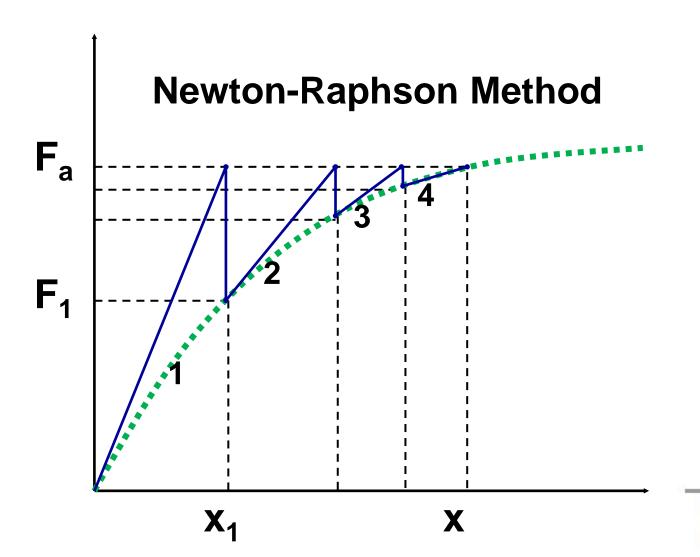
 $\begin{bmatrix} M \end{bmatrix}$ ,  $\begin{bmatrix} C \end{bmatrix}$ ,  $\begin{bmatrix} K(\mathsf{t}) \end{bmatrix}$  symmetrical







#### Linear and **Nonlinear** Dynamic Analysis

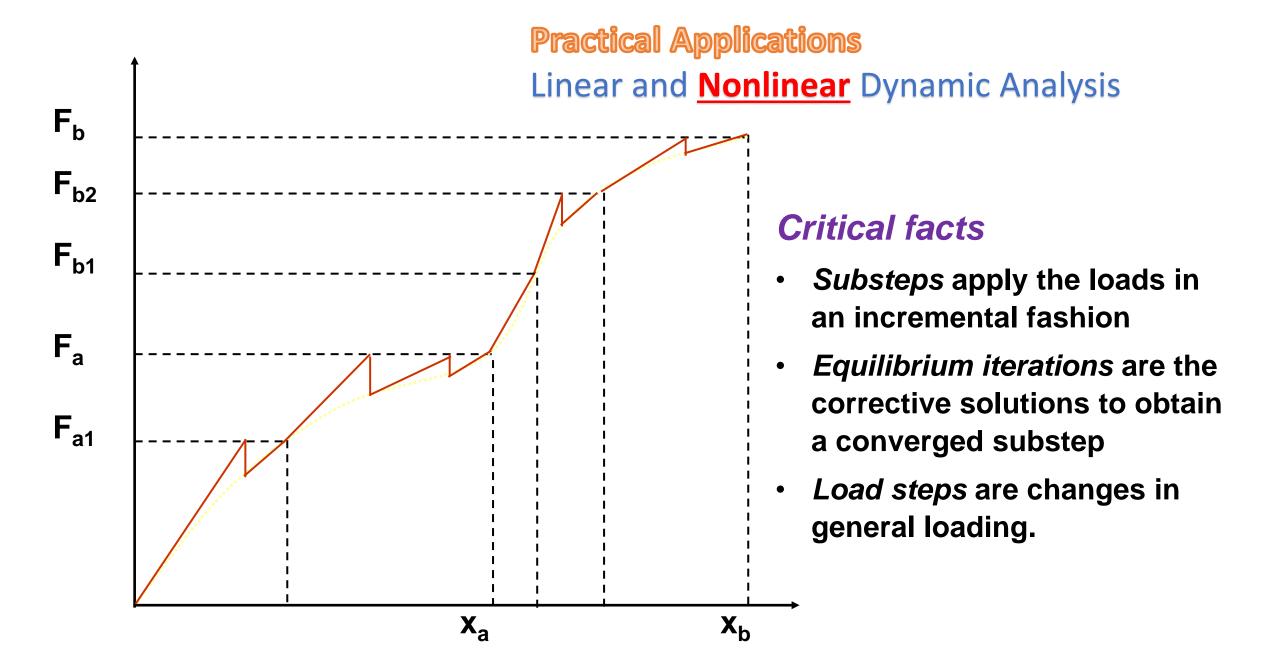


- The actual relationship between load and displacement is not known beforehand.
- Consequently, a series of linear approximations with corrections is performed.
   This is a simplified explanation of the Newton-Raphson method.



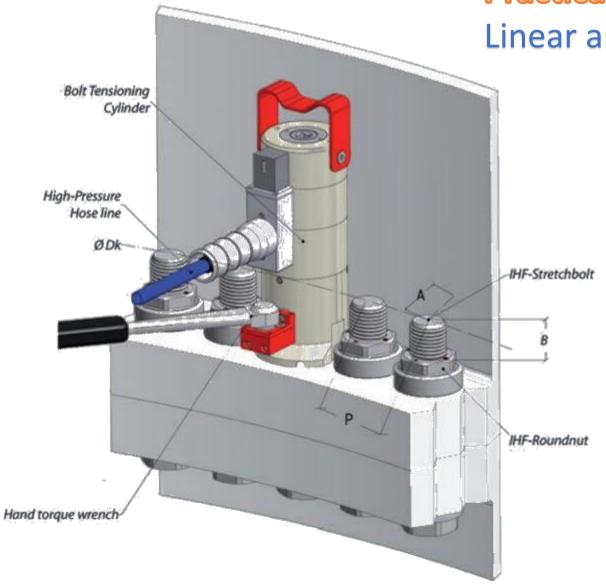








Linear and **Nonlinear** Dynamic Analysis



#### Critical facts

- Substeps apply the loads in an incremental fashion
- Equilibrium iterations are the corrective solutions to obtain a converged substep
- Load steps are changes in general loading.

Linear and **Nonlinear** Dynamic Analysis

The <u>complete</u> <u>nonlinear</u> dynamic aeroelastic (or aeroinelastic) problem

$$[M] \cdot \ddot{u} + [C+G] \cdot \dot{u} + [K(t)+Z(t)+H(t)] \cdot u = R(t)$$

Mass matrix

C Damping matrix

G Coriolis matrix

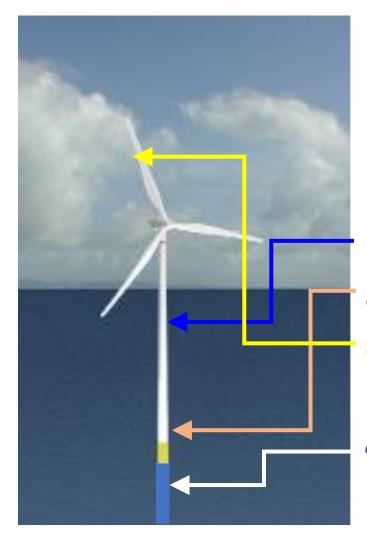
 $egin{bmatrix} K(t) \end{bmatrix}$  Stiffness matrix  $egin{bmatrix} M \end{bmatrix}$  ,  $egin{bmatrix} C \end{bmatrix}$  ,  $egin{bmatrix} K \end{bmatrix}$  and  $egin{bmatrix} Z \end{bmatrix}$  symmetrical

[Z(t)] Centrifugal stiffness matrix [G] and [H] anti-symmetrical

H(t) Acceleration stiffness matrix

R(t) Vector of external loads

 $u, \dot{u}, \ddot{u}$  Displacement-, velocity- and acceleration vectors

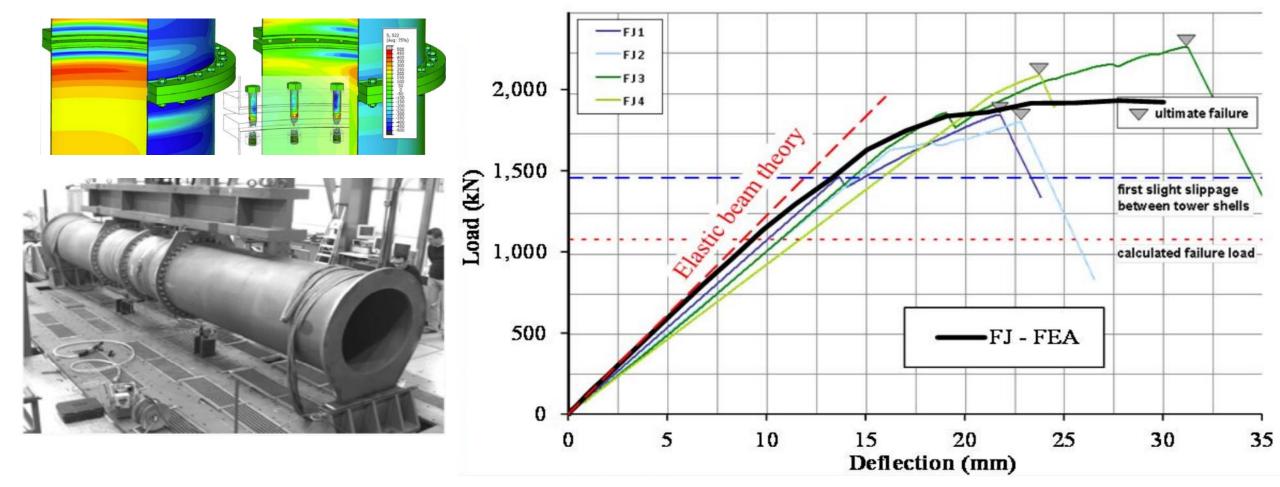


## NONLINEAR STRUCTURAL BEHAVIOR OF WIND TURBINE COMPONENTS

#### **CASE STUDIES**

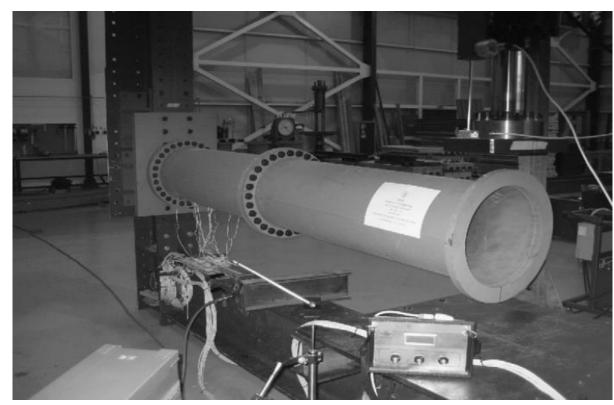
- 1) Ring flange connections
- 2) Stability of tower (opening/stiffening)
- 3) Delamination & local buckling in blades
- 4) Nonlinear behavior of monopile

## Nonlinear structural engineering and wind turbines Case study 1: Ring flange connection



Veljkovic, M. et al (2010), 'High-strength steel tower for wind turbine, HISTWIN' Final Report RFSR-CT-2006-00031, Brussels: RFCS Publications, European Commission

## Nonlinear structural engineering and wind turbines Case study 2: Nonlinear behavior of wind turbine towers



C.A. Dimopoulos, C.J. Gantes, Experimental investigation of buckling of wind turbine tower cylindrical shells with opening and stiffening under bending, Thin-Walled Structures, vol. 54, pp. 140-155, 2012.



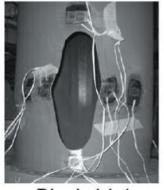




Block 12-1

Block 12-2

**GMNA** 







Block 14-1

Block 14-2

**GMNA** 





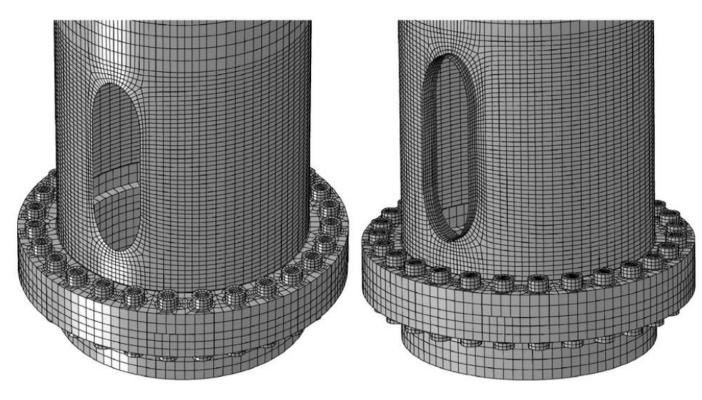


Block 16-1

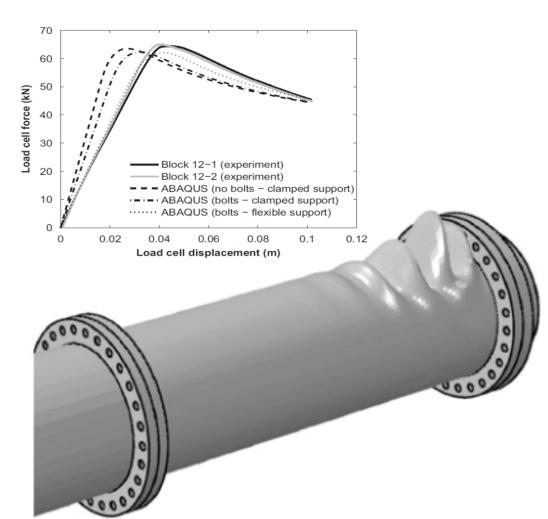
Block 16-2

**GMNA** 

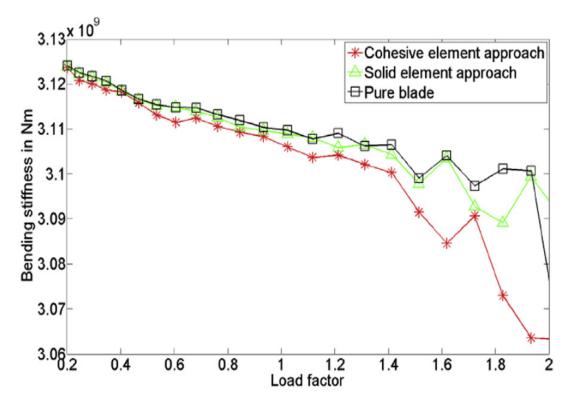
## Nonlinear structural engineering and wind turbines Case study 2: Nonlinear behavior of wind turbine towers



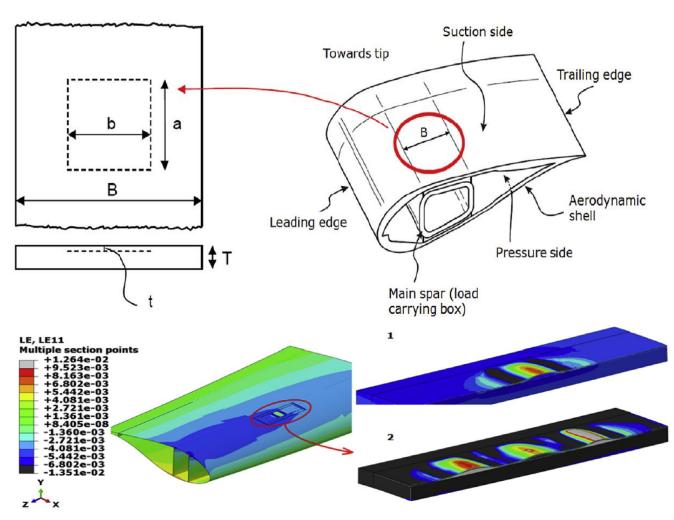
C.A. Dimopoulos, C.J. Gantes, Experimental investigation of buckling of wind turbine tower cylindrical shells with opening and stiffening under bending, Thin-Walled Structures, vol. 54, pp. 140-155, 2012.



## Nonlinear structural engineering and wind turbines Case study 3: Nonlinear behavior of wind turbine blades

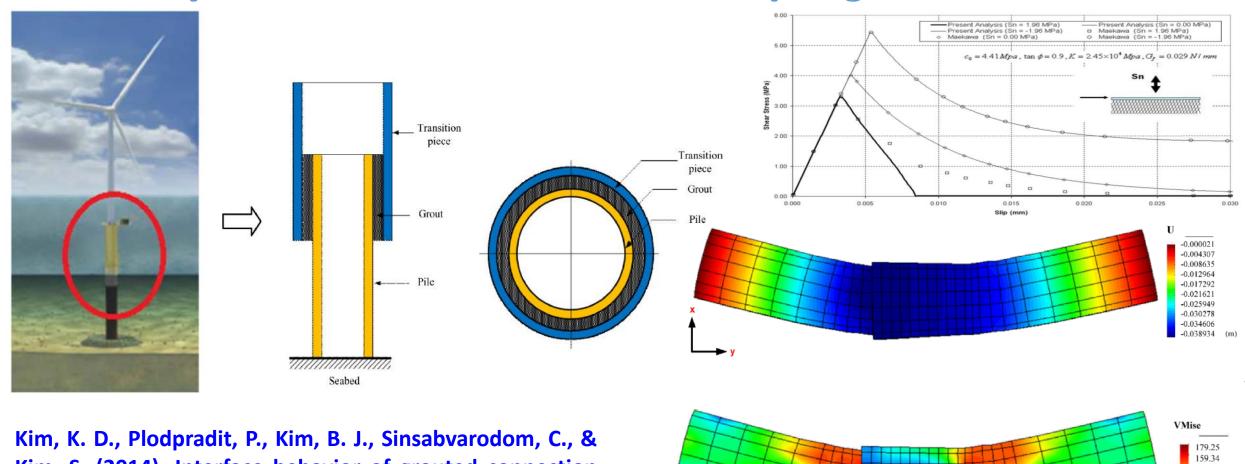


Haselbach, P. U., Bitsche, R. D., & Branner, K. (2016). The effect of delaminations on local buckling in wind turbine blades. Renewable Energy, 85, 295-305.

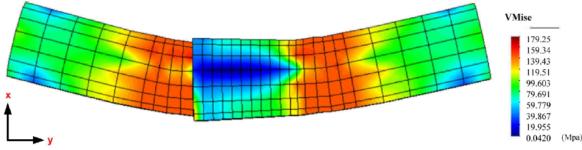


## Nonlinear structural engineering and wind turbines

Case study 4: Nonlinear behavior of monopile grouted connection



Kim, K. D., Plodpradit, P., Kim, B. J., Sinsabvarodom, C., & Kim, S. (2014). Interface behavior of grouted connection on monopile wind turbine offshore structure. International Journal of Steel Structures, 14(3), 439-446.



# Thank you for your attention



